



ESA's Rosetta mission on the comet 67P/Churyumov-Gerasimenko

M. Fulchignoni





A comet rendezvous and two asteroid flybys mission

Rosetta

Flying through and landing on a comet

ESA Rosetta mission

- First rendezvous to a comet Ambitious mission, ESA cornerstone aimed at the deciphering of our origins
- Scientific objectives
- Characterize the evolution of activity from aphelion to perihelion
- Determine the nucleus properties (structure, chemical & isotopic composition, physical & mineralogical properties), by orbiter payload & landing a modular laboratory on the still mysterious nucleus surface



Solar panels: 32m x1m



Flying through and landing on a comet

ESA Rosetta mission

- Constraints related to very low gravity field of comet nuclei
 ⇒ Relative velocity ≈ 0 at encounter
- Constraints on the trajectory
- \Rightarrow Trajectory parallel to that of the nucleus or around it for \approx 1 yrs
- ⇒ Encounter near aphelion (smallest orbital velocity)
- ⇒ Near ecliptic trajectory with perihelion near 1 AU (less demanding)
- ⇒ Aphelion not too far away (Jupiter family comet)

Building Rosetta

Massive and complex spacecraft

- Long duration mission
- Long cruise phase (gravity assists and trajectory corrections)
- Hibernation phase
- Long rendezvous phase (2 years +)
- About 10 sophisticated experiments on the spacecraft, as well as on the "Philae" lander

⇒ About 3 m x 2 m x 2 m
three axis stabilized spacecraft
⇒ 1.3 t (dry mass) + 1.6 t (fuel)
(Philae = 110 kg)

Integration the high gain antenna on a first model



Rosetta Spacecraft and Payload



Rosetta Spacecraft and Payload



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Philae Lander and Payload

Innovative small instruments for Philae

Imaging Composition analysis Physical properties Nucleus large-scale structure Magnetic field and plasma Drill and sampling device CIVA, ROLIS APX, COSAC, Ptolemy MUPUS, SESAME CONSERT ROLIS SD2

The Rosetta challenges

- Rendezvous with a low gravity object
- Landing on a mysterious object (unknown porosity, variable activity)
- Survey from 50 km (surface mapping) down to 1 km
- Descent of the probe and deployment of the landing gear with tripod, anchors and harpoons



ESA artist views of the mission



Temperature and solar energy constraints

Sophisticated solar panels
32 m long x 1m large , 25 % efficiency



Journey to Churyumov-Gerasimenko

2003-2004, expecting the launch...

- Spacecraft fully integrated, flown to Kourou in September 2003
- Report of the launch to comet Wirtanen (anticipated to take place in January 2003) after an Ariane 5 failure





The new target

• Churyumov-Gerasimenko Orbital parameters $a = 3.507 \text{ UA} \quad i = 7.12^{\circ} \quad e = 0.632$ $a^{3}/P^{2} = 1 \text{ UA}^{3} \text{ year}^{-2} \quad \Rightarrow P = 6.57 \text{ years}$ $p = a(1-e) \quad \Rightarrow p = 1.29 \text{ UA}$ $p' = a(1+e) \quad \Rightarrow p' = 5.74 \text{ UA}$

 Physical properties (estimated from HST observations)
 Nucleus about 3 km x 5 km





Journey to Churyumov-Gerasimenko

Launch by Ariane 5G+

- One day report (meteorological and three more days report (insulation improvement)
- Successful launch, 2 March 2004
- After about 2 hours and 3 crossings of the radiation belts, successful upper stage ignition for interplanetary injection



Journey to Churyumov-Gerasimenko

• All commissioning satisfactory

• Images from Osiris Camera of comet C/2002 T7 LINEAR



Rosetta was also seen from the Earth



Deep impact observations in July 2005

Before impact After impact **ROSETTA -- OSIRIS NAC** (C)2005 MPS for OSIRIS Team MPS/UPD/LAM/IAA/RSSD/INTA/UPM/DASP/IDA - ESA



ROSETTA ON ITS WAY TO **67P/Churyumov-Gerasimenko IS VISITING** STEINS AND LUTETIA ASTEROIDS

January 2014 Exit deep space hibernation

May 2014 Major rendezvous manoeuvre

August 2014 Arrive at comet

November 2014 Lander delivery

August 2015 Closest approach to Sun







→ COMET 67P/CHURYUMOV-GERASIMENKO'S VITAL STATISTICS



21.4 km³ Volume 1.0 × 1013 kg Mass

> 470 kg/m³ Density

70-80% Porosity





Dust/gas ratio

5.3 × 10-4 D/H ratio

Average water vapour production 0-300 ml/s + June 2014 600 ml/s → July 2014 1200 ml/s > August 2014

3.3 km





64.1° Declination

Spin axis:

Rotation period

69.3° Right Ascension

12.4043 hours

52° Obliquity of the comet's rotational axis



XY Equatorial axes Z Spin axis

-93°C to -43°C Surface temperature

-243°C to -113°C Subsurface temperature

6% Average albedo

te: MIRO; D/H: ROSINA GIADA MIS t images: NavCam

European Space Agency

→ PHILAE'S LANDING SITE







Agilkia region

Les dix instruments de mesures de Philae

CONSERT

Sondeur du noyau cométaire. Réalise un scanner en 3 dimensions du cœur de la comète.

COSAC et PTOLEMY Analyseurs de gaz. Étudient la composition chimique des échantillons prélevés sous la surface.

SESAME Sondeur électrique et acoustique. Mesure les propriétés mécaniques et électriques de la surface et étudie les poussières émises par la comète.

SD2 Collecteur d'échantillons. Fore sous la surface pour prélever des échantillons et les distribuer à 3 autres instruments (Çiva, Cosac, Ptolemy). ROMAP Magnétomètre et • analyseur de plasma. Étudie le champ magnétique et les ondes plasma émises par la surface.

CIVA / ROLIS Système d'imagerie. Sept caméras réalisent des images panoramiques 3D. Une caméra observe la morphologie de la surface. Un microscope analyse les échantillons prélevés sous la surface.

 APXS Spectromètre à rayons X des particules alpha. Analyse la composition chimique du site d'atterrissage.

MUPUS Ensemble de capteurs. Mesure la dureté du sol et sa température.

Crédit: ESA/Rosetta/Philae/CIVA



Agilkia region



Gravitational Slopes



Geomorphological units

- diamicton deposit
 - gravitational accumulation deposit
- talus deposit
- fine material deposit on layered terrain outcropping layered terrain

Linear features



200 m





NAVCAM 15:30-15:34

1. 250m

Rosetta @15km

Res=1.5 m





Touchdown 15:34





CONSERT estimation of landing area

Additional candidate assuming shape model deviation

350 m

3011



Landing area based on current shape model

38





Superimposed on top of the image is a sketch of the Philae lander in the configuration the lander team currently believe it is in.





CNES reconstruction





Close flyby



February 14, 2015 at 6 km (Res.= 11 cm) Activity everywhere



